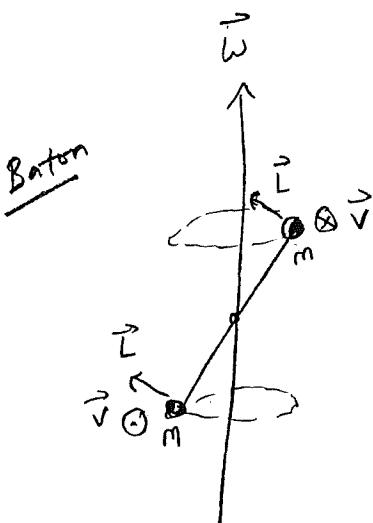


Q1

Unbalanced rigid body  $\Rightarrow \vec{L}$  not parallel to  $\vec{\omega}$



$$\vec{L} = \sum m_i \vec{r}_i \times \vec{v}_i$$

$$\text{Recall } \vec{v} = \vec{\omega} \times \vec{r}$$

$$\text{Let } \vec{\omega} = \omega_z \hat{z} = \begin{pmatrix} 0 \\ 0 \\ \omega \end{pmatrix}$$

$$\vec{r} = x \hat{x} + y \hat{y} + z \hat{z} = \begin{pmatrix} x \\ y \\ z \end{pmatrix}$$

[Use  $\hat{x}, \hat{y}, \hat{z}$   
not  $\hat{i}, \hat{j}, \hat{k}$   
so as not to  
confuse w/  
indication]

$$\vec{v} = \begin{vmatrix} \hat{x} & \hat{y} & \hat{z} \\ 0 & 0 & \omega \\ x & y & z \end{vmatrix} = \begin{pmatrix} -\omega y \\ \omega x \\ 0 \end{pmatrix}$$

$$\vec{r} \times \vec{v} = \begin{vmatrix} \hat{x} & \hat{y} & \hat{z} \\ x & y & z \\ -wy & wx & 0 \end{vmatrix} = \begin{pmatrix} -wyz \\ -wzx \\ wx^2 + wy^2 \end{pmatrix}$$

$$\vec{L} = \sum m_i \vec{r}_i \times \vec{v}_i = \begin{pmatrix} \sum (-m_i x_i z_i) \\ \sum (-m_i y_i z_i) \\ \sum m_i (x_i^2 + y_i^2) \end{pmatrix} \omega_z \equiv \begin{pmatrix} I_{xz} \\ I_{yz} \\ I_{zz} \end{pmatrix} \omega_z$$

$I_{xz}, I_{yz}$  = products of inertia

$I_{zz}$  = moment of inertia

$$L_x = I_{xz} \omega$$

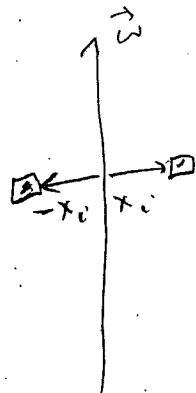
$$L_y = I_{yz} \omega$$

$$L_z = I_{zz} \omega$$

$\vec{L} \parallel \vec{\omega}$  iff products of inertia vanish

Q2

If object is balanced wrt. z-axis



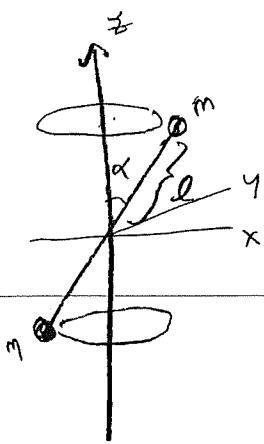
$$\sum m_i x_i z_i = 0 \Rightarrow I_{xz} = 0$$

$$\sum m_i y_i z_i = 0 \Rightarrow I_{yz} = 0$$

$$\Rightarrow \vec{T} = I_{zz} \hat{z} \omega = I_{zz} \vec{\omega}$$

$$I_{zz} = \sum m_i (x_i^2 + y_i^2) = \sum m_i r_i^2$$

as before



Assume  $\vec{\omega} = \omega \hat{z}$

with  $\omega = \text{const}$

$$\begin{cases} x = l \sin \alpha \cos \omega t \\ y = l \sin \alpha \sin \omega t \\ z = l \cos \alpha \end{cases}$$

$$I_{xz} = -2ml^2 \sin \alpha \cos \alpha \cos \omega t$$

$$I_{yz} = -2ml^2 \sin \alpha \cos \alpha \sin \omega t$$

$$I_{zz} = -2ml^2 \sin^2 \alpha$$

N.B.  $I_{\dots}$  depend on time (in inertial ref frame)

$$\vec{L} = \begin{pmatrix} I_{xz} \\ I_{yz} \\ I_{zz} \end{pmatrix} \vec{\omega} = 2ml^2 \omega \begin{pmatrix} -\cos \alpha \cos \omega t \\ -\cos \alpha \sin \omega t \\ \sin \alpha \end{pmatrix}$$

$\vec{L}$  not conserved  $\Rightarrow$  torque required to maintain rotation

$$\vec{\tau} = \frac{d\vec{L}}{dt} = 2ml^2 \omega \begin{pmatrix} \cos \alpha \sin \omega t \\ -\cos \alpha \cos \omega t \\ 0 \end{pmatrix}$$



These forces supply the necessary torque

[check  $\vec{r} \times \vec{F}$  out of page]

"fighting centrifugal force"

[unbalanced washing machine]

Q4

Rotation about an arbitrary axis:  $\vec{\omega} = \begin{pmatrix} \omega_x \\ \omega_y \\ \omega_z \end{pmatrix}$

For an arbitrary mass distribution

$$\vec{L} = \int dm \vec{r} \times \vec{v}$$

$$= \int dm \vec{r} \times (\vec{\omega} \times \vec{r})$$

Identity  $\vec{a} \times (\vec{b} \times \vec{c}) = (\vec{a} \cdot \vec{c})\vec{b} - (\vec{a} \cdot \vec{b})\vec{c}$

outer dot remote times adjacent  
minus outer dot adjacent times remote

$$\vec{L} = \int dm [ r^2 \vec{\omega} - (\vec{r} \cdot \vec{\omega}) \vec{r} ]$$

$$= \int dm [ (x^2 + y^2 + z^2) \begin{pmatrix} \omega_x \\ \omega_y \\ \omega_z \end{pmatrix} - (x\omega_x + y\omega_y + z\omega_z) \begin{pmatrix} x \\ y \\ z \end{pmatrix} ]$$

$$= \int dm \begin{pmatrix} (y^2 + z^2)\omega_x - xy\omega_y - xz\omega_z \\ -xy\omega_x + (x^2 + z^2)\omega_z - yz\omega_z \\ -xz\omega_x - yz\omega_y + (x^2 + y^2)\omega_z \end{pmatrix}$$

$$= \int dm \begin{pmatrix} y^2 + z^2 & -xy & -xz \\ -xy & x^2 + z^2 & -yz \\ -xz & -yz & x^2 + y^2 \end{pmatrix} \begin{pmatrix} \omega_x \\ \omega_y \\ \omega_z \end{pmatrix}$$

$$\begin{pmatrix} L_x \\ L_y \\ L_z \end{pmatrix} = \begin{pmatrix} I_{xx} & I_{xy} & I_{xz} \\ I_{yx} & I_{yy} & I_{yz} \\ I_{zx} & \underbrace{I_{zy}} & I_{zz} \end{pmatrix} \begin{pmatrix} \omega_x \\ \omega_y \\ \omega_z \end{pmatrix}$$

Note that  
matrix is  
symmetric  
 $I^T = I$

Redo using index notation:  $v_m$  ( $m=1, 2, 3$ ) are Cartesian components of  $\vec{v}$

$$\begin{aligned} L_m &= \int dm \left[ r^2 \omega_m - (\vec{r} \cdot \vec{\omega}) r_m \right] \\ &= \int dm \left[ \omega_m \sum r_i^2 - r_m \sum r_n \omega_n \right] \\ &= \int dm \sum_n \left[ \delta_{mn} \sum r_i^2 - r_m r_n \right] \omega_n \end{aligned}$$

Recall  
 $\delta_{mn}$  = Kronecker delta  
 $= \begin{cases} 0 & m \neq n \\ 1 & m = n \end{cases}$

$$L_m = \sum_n I_{mn} \omega_n$$

$$\sum \delta_{mn} \omega_n = \omega_m$$

$$\text{where } I_{mn} = \int dm \left[ \delta_{mn} \left( \sum r_i^2 \right) - r_m r_n \right]$$

(Note that  $I_{mn} = I_{nm} \Rightarrow$  symmetric matrix)  $\xrightarrow{\text{does not} \rightarrow \text{Hermitian}}$

Diagonal elements  $I_{mm} = \int dm \left[ \left( \sum r_i^2 \right) - r_m^2 \right] = \int dm r_{\perp m}^2 = \frac{\text{moment of inertia}}{\text{mass}}$

off-diagonal elements  $I_{mn} = - \int dm r_m r_n = \text{products of inertia}$

Another notation

$$\vec{L} = \overset{\leftrightarrow}{I} \cdot \vec{\omega}$$

$\uparrow$   
"dyad" acts on vector to produce a vector in a (possibly) different direction

QF

Kinetic energy

$$T = \frac{1}{2} \int dm v^2$$

$$= \frac{1}{2} \int dm \vec{v} \cdot (\vec{\omega} \times \vec{r})$$

$$= \frac{1}{2} \int dm \vec{\omega} \cdot (\vec{r} \times \vec{v})$$

$$= \frac{1}{2} \vec{\omega} \cdot \left( \int dm (\vec{r} \times \vec{v}) \right)$$

$$= \frac{1}{2} \vec{\omega} \cdot \vec{I}$$

$$= \frac{1}{2} \vec{\omega} \cdot \overset{\leftrightarrow}{I} \cdot \vec{\omega}$$

$$T = \frac{1}{2} \sum_m w_m I_{mm} \omega_m^2$$

If rotating around  $\hat{z} = \hat{x}i$ , reduce to

$$T = \frac{1}{2} I_{zz} \omega^2 = \frac{1}{2} \left( \int dm (r_{\hat{z}}^2) \right) \omega^2$$

$\left\{ \begin{array}{l} \text{Treated about} \\ \text{less count} \\ r = r_{\hat{x}\hat{z}} \\ q_i = \sum m_i (r_{\hat{x}\hat{z}})^2 \end{array} \right.$

## Principal axes

If  $I$  is a diagonal matrix, i.e.  $\overset{I}{\leftarrow} = \begin{pmatrix} I_{xx} & 0 & 0 \\ 0 & I_{yy} & 0 \\ 0 & 0 & I_{zz} \end{pmatrix}$   
 the object is balanced wrt.  $\hat{x}, \hat{y}, \hat{z}$  axes

---

If  $I_{xy} \neq 0$  object is not balanced wrt. wrt.  $\hat{x} + \hat{y}$  axes  
 and similarly for  $I_{xz} + I_{yz}$ .

---

However, any object, however misshapen, is always balanced wrt at least 3 axes  $\hat{e}^1, \hat{e}^2, \hat{e}^3$   
 called principal axes  
 (superscripts label 3 different unit vectors)  
 (subscript = components)

The moments of inertia wrt. these axes,  $I_1, I_2, I_3$   
 are called principal moments.

Thus if  
 $\vec{\omega} = \omega \hat{e}^k$  for  $k=1, 2, 3$ , then  $\vec{L} = I_k \vec{\omega}$

---

Furthermore, the principal axes are mutually perpendicular

How do we find these principal axes and moments?

In general

$$\vec{L} = \overset{\leftrightarrow}{I} \cdot \vec{\omega}$$

If  $\vec{\omega}$  is parallel to a principal axis  $\hat{e}$  then

$$\vec{L} = I \vec{\omega}$$

Thus

$$\overset{\leftrightarrow}{I} \cdot \vec{\omega} = I \vec{\omega}$$

in components

$$\sum_n I_{mn} e_n = I e_m$$

$$\therefore \overset{\leftrightarrow}{I} \cdot \hat{e} = I \hat{e}$$

Eigenvalue eqn w eigenvalues  $I$  and eigenvec  $\hat{e}$ .

Write it as

$$[\overset{\leftrightarrow}{I} - I \overset{\leftrightarrow}{I}] \cdot \hat{e} = 0 \quad \text{where } \overset{\leftrightarrow}{I} = \begin{matrix} \text{identity matrix} \\ (1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1) \end{matrix}$$

or

$$\begin{pmatrix} I_{xx} - I & I_{xy} & I_{xz} \\ I_{yx} & I_{yy} - I & I_{yz} \\ I_{zx} & I_{zy} & I_{zz} - I \end{pmatrix} \begin{pmatrix} \hat{e}_x \\ \hat{e}_y \\ \hat{e}_z \end{pmatrix} = 0$$

This has nontrivial solutions for  $\hat{e}$  if

$$\det(\overset{\leftrightarrow}{I} - I \overset{\leftrightarrow}{I}) = 0$$

This is a polynomial cubic in  $I$

Solve the cubic eqn for 3 values of  $I_j$   
 call them  $I_1, I_2, I_3$

(In principle, some of the roots could be complex  
 but for a symmetric matrix, they are all real.)

For each eigenvalue  $I_k$ , solve for the  
 corresponding eigenvector  $\hat{e}^k$ .

Eigenvalue equation only determines the direction,  
 not magnitude, of  $\hat{e}^k$ , so we normalize it  
 to have length 1.

[ $\hat{e}^k$  are uniquely determined  
 only if eigenvalues are non-degenerate]

$$\sum_n I_{mn} \hat{e}_n^k = I_k \hat{e}_m^k$$

RY

Proof that eigenvectors are mutually orthogonal.

Let  $\hat{e}^k$  and  $\hat{e}^l$  be eigenvectors of distinct eigenvalues ( $I_k \neq I_l$ )

$$\overset{\leftrightarrow}{I} \cdot \hat{e}^k = I_k \hat{e}^k$$

$$\overset{\leftrightarrow}{I} \cdot \hat{e}^l = I_l \hat{e}^l$$

Dot w/  $\hat{e}^k$

Dot w/  $\hat{e}^l$

$$(1) \quad \hat{e}^l \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^k = I_k \hat{e}^l \cdot \hat{e}^k \quad (2) \quad \hat{e}^k \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^l = I_l \hat{e}^k \cdot \hat{e}^l$$

$$\text{Claim: } \hat{e}^l \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^k = \hat{e}^k \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^l \quad (3)$$

$$\text{proof: } \sum_{m,n} \hat{e}_m^l I_{mn} \hat{e}_n^k = \sum_{m,n} \hat{e}_m^l I_{pm} \hat{e}_{pn}^k = \sum_{m,n} \hat{e}_m^k I_{pn} \hat{e}_n^l$$

↑                          ↑  
I is symmetric            rearrange

$$= \sum_m \hat{e}_m^k I_{mn} \hat{e}_n^l = \hat{e}^k \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^l \quad \text{QED}$$

relabel m  $\leftrightarrow$  n

Putting (1), (2), (3) together

$$I_k \hat{e}^l \cdot \hat{e}^k = I_l \hat{e}^k \cdot \hat{e}^l$$

$$\Rightarrow (I_k - I_l) \hat{e}^l \cdot \hat{e}^k = 0$$

$$\text{But } I_k \neq I_l \text{ so } \hat{e}^l \cdot \hat{e}^k = 0 \quad \text{QED}$$

## Proof of reality of eigenvalue

$$\overset{\leftrightarrow}{I} \cdot \hat{e}^k = I_k \hat{e}^k$$

Dot w/  $(\hat{e}^k)^*$

$$\hat{e}^{k*} \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^k = I_k \hat{e}^{k*} \cdot \hat{e}^k \quad (1)$$

Take c.c. and we find that  $\overset{\leftrightarrow}{I}$  is real

$$\hat{e}^k \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^{k*} = I_k^* \hat{e}^k \cdot \hat{e}^{k*} \quad (2)$$

$$\hat{e}^k \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^{k*} = \hat{e}^k \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^{k*} = \hat{e}^{k*} \cdot \overset{\leftrightarrow}{I} \cdot \hat{e}^k \quad (3)$$

But by previous calc:

Putting (1), (2), (3) together

$$I_k \hat{e}^{k*} \cdot \hat{e}^k = I_k^* \hat{e}^{k*} \cdot \hat{e}^k$$

$$(I_k - I_k^*) \hat{e}^{k*} \cdot \hat{e}^k = 0$$

$$\text{But } \hat{e}^{k*} \cdot \hat{e}^k = \sum |e_i^k|^2 > 0$$

$$\text{so } I_k - I_k^* = 0$$

$$I_k^* = I_k$$

QED

omit this

unless someone  
questions it

Note we  
should  
also show  
reality of  
eigenvectors  
(ie the all  
the components  
can be  
shown  
real)

(3)

only needs  
it to work  
but not  
real

RS

(12-13-23)

⑨

Rotations

N.B.  $R$  here is the transpose of my adv. mech. notes, if  $\hat{e}_n = \hat{x}^n$   $\rightarrow$  space

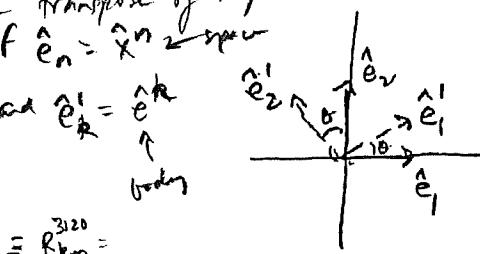
Let  $R$  rotate basis  $\hat{e}_n$  into  $\hat{e}'_n$  and  $\hat{e}'_k = \hat{e}^k$   
 $\hat{e}^k = R \hat{x}^k$  body

$$\hat{e}'_n = R \hat{e}_n \Rightarrow R_{mk} = \hat{x}^m \cdot \hat{e}^k = e^k_m \equiv R^{320}$$

$$\text{Define } R_{mn} \equiv \hat{e}_m \cdot \hat{e}'_n = \hat{e}_m \cdot \hat{e}_n$$

$$\hat{e}'_n = \sum_m \hat{e}_m \cdot \hat{e}_m \cdot R \hat{e}_n = \sum_m \hat{e}_m R_{mn} \quad \text{N.B. Thus } \hat{e}'_n = R^T \hat{e}_m$$

$$R = \sum_{m,n} \hat{e}_m R_{mn} \hat{e}_n$$



$$R_{21} = \hat{e}_2 \cdot \hat{e}'_1 = \sin \theta$$

$$R_{12} = \hat{e}_1 \cdot \hat{e}'_2 = -\sin \theta$$

$$R = \begin{pmatrix} c & -s \\ s & c \end{pmatrix}$$

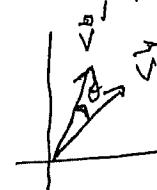
Active rotation of  $\vec{v}$ 

$$\text{Define } v_n = \hat{e}_n \vec{v} \Rightarrow \vec{v} = \sum_n \hat{e}_n v_n$$

$$\vec{v}' = R \vec{v} = \sum_n (R \hat{e}_n) v_n = \sum_n \hat{e}'_n v_n \quad \text{compare}$$

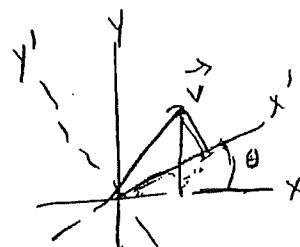
$$\text{Define } \boxed{v'_n = \hat{e}_n \vec{v}' \Rightarrow \vec{v}' = \sum_n \hat{e}_n v'_n}$$

$$v'_n = \hat{e}_n \cdot R \vec{v} = \sum_m \hat{e}_n \cdot R \hat{e}_m \hat{e}_m \vec{v} = \sum_m R_{nm} v_m$$

Passive rotation

Let  $\vec{v}$  be fixed and define

$$\boxed{v_n = \hat{e}_n \cdot \vec{v}, \quad v'_n = \hat{e}'_n \cdot \vec{v}} \quad (\text{differs from above})$$



$$v'_n = \sum_m \hat{e}'_n \cdot \hat{e}_m \hat{e}_m \vec{v} = \sum_m R_{mn} v_m = \sum_m (R^T)_{nm} v_m \quad R^T = \begin{pmatrix} c & s \\ -s & c \end{pmatrix}$$

(1-19 - 24)

Background material  
I worked this out for my own satisfaction!

P3120  
Spr 2024

Let  $R$  be a real, orthogonal matrix s.t.  $R^T R = I$

Consider its eigenvalues  $Rv = \lambda v$  for non-zero  $v$

$$v^T R^T = \lambda^* v^T$$

$$\underbrace{v^T R^T R}_I v = |\lambda|^2 v^T v$$

$$\underbrace{R^T R}_I$$

Since  $v^T v$  is manifestly nonzero (if  $v$  is non-zero),  $\lambda = e^{i\theta}$

(Observe that  $v^T R^T = \lambda v^T \Rightarrow v^T v = \lambda^2 v^T v$

so either  $\lambda = \pm 1$ , or  $v^T v = 0$ .)

Restrict now to  $3 \times 3$  matrices.

Let  $\hat{n}$  be the eigenvector of  $R$  of eigenvalue 1

$$R \hat{n} = \hat{n}$$

- if  $\hat{n} \cdot \vec{L}$  NB. This corresponds to an active rotation through  $\phi$  (passive would have opp sign)

$$\text{Then } R = e^{\vec{L}\phi}$$

$$\text{where } (\vec{L}^a)_{mn} = i E^{\text{man}}$$

This is because  $(\underbrace{\hat{n} \cdot \vec{L}}_{\text{in } E^{\text{man}}}) \hat{n}_{ii} = 0 \Rightarrow R \hat{n} = \hat{n}$

$\hat{n}_{ii}$

The often two eigenvectors are complex, & eigenvalues are  $e^{\pm i\phi}$

$$\text{e.g. } \begin{pmatrix} \cos \phi & -\sin \phi & 0 \\ \sin \phi & \cos \phi & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} 1 \\ i \\ 0 \end{pmatrix} = e^{\mp i\phi} \begin{pmatrix} 1 \\ i \\ 0 \end{pmatrix}$$

## Background

$$R^T R = I$$

$$R = e^r \Rightarrow r^T = r$$

$R$  can be expressed as a rotation about  $\vec{v}$  through angle  $\theta$

$R v = v$  solve for  $v$  to find axis of rotation (normalize it)

$$\det(R - \lambda I) = 0 \Rightarrow \lambda = 1, e^{i\theta}, e^{-i\theta} \text{ for angle of rotation}$$

$$\text{Then } r_{mn} = \theta E_{mn} v_l \quad (\exists \text{ sign ambiguity in b/c } \theta + v)$$

$$\text{Eg. } R = \begin{pmatrix} \cos \theta & \sin \theta & 0 \\ -\sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

$$Rv = v \Rightarrow v = \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix}$$

$$[(\lambda - \cos \theta)^2 + \sin^2 \theta][\lambda - 1] = 0 \quad \lambda = e^{i\theta}, e^{-i\theta}, 1$$

$$r = \begin{pmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{pmatrix}$$

Background

Fetter-Walecka: Theoretical mechanics, §7 (p.33)

Tong: Classical Dynamics, §3

Let  $\hat{e}^k$  be orthonormal body-fixed axes:  $\hat{e}^k \cdot \hat{e}^l = \delta_{kl}$

Goal: show  $\frac{d\hat{e}^k}{dt} = \vec{\omega} \times \hat{e}^k$

Proof:

$$\text{Let } \frac{d\hat{e}^k}{dt} = \omega_{kl} \hat{e}^l \Rightarrow \omega_{kl} = \frac{d\hat{e}^k}{dt} \cdot \hat{e}^l$$

$$\text{orthonormality } \Rightarrow \omega_{kl} = -\omega_{lk}$$

$$\text{Define } \omega_{kl} = \epsilon_{klm} \omega_m \Rightarrow \frac{d\hat{e}^k}{dt} = \epsilon_{klm} \omega_m \hat{e}^l$$

$$\omega_m = \frac{1}{2} \epsilon_{mkl} \omega_{kl} = \frac{1}{2} \epsilon_{mkl} \frac{d\hat{e}^k}{dt} \cdot \hat{e}^l$$

Define  $\vec{\omega} = \omega_m \hat{e}^m$  as  $\omega_m$  = body-fixed components of  $\vec{\omega}$

$$\text{Then } \vec{\omega} \times \hat{e}^k = \omega_m \hat{e}^m \times \hat{e}^k = \omega_m \epsilon_{mkl} \hat{e}^l = \frac{d\hat{e}^k}{dt} \quad (\text{QED})$$

Note  $R_{ki} = \hat{e}_i^k$

$$R_{ki} R_{li} = \delta_{kl}$$

(cf Tong)  
§3.1.1

$$\omega_{kl} = \left( \frac{d}{dt} \hat{e}_i^k \right) \hat{e}_i^l = \frac{dR_{ki}}{dt} R_{li} = -R_{ki} \frac{dR_{li}}{dt} = -\omega_{kl}$$

$$\left( \frac{d\vec{A}}{dt} \right)_{\text{spac}} = \sum \frac{dA_i}{dt} \hat{x}_i = \sum \left( \frac{dA_k'}{dt} \hat{e}_k + A_k' \frac{d\hat{e}_k}{dt} \right)$$

$$= \sum \left( \frac{dA_k'}{dt} \right) \hat{e}_k + \vec{\omega} \times \sum A_k' \hat{e}_k$$

$$\boxed{\left( \frac{d\vec{A}}{dt} \right) = \left( \frac{d\vec{R}}{dt} \right)_+ + \vec{\omega} \times \vec{A}}$$

## Background

$$\hat{e}^k = R_{km} \hat{x}^m \Rightarrow \hat{x}^m = (R^{-1})_{ml} \hat{e}^l$$

$$\begin{aligned}\frac{d\hat{e}^k}{dt} &= \dot{R}_{km} \hat{x}^m = \dot{R}_{km} (R^{-1})_{ml} \hat{e}^l \\ &= (\dot{R} R^{-1})_{kl} \hat{e}^l\end{aligned}$$

Define  $w_{kl} \equiv (\dot{R} R^{-1})_{kl} = (\dot{R} R^T)_{kl} \Rightarrow w^T = R \dot{R}^T$

$$RR^T = I \Rightarrow \dot{R} R^T + R \dot{R}^T = 0$$

$$w + w^T = 0 \Rightarrow w_{kl} = \epsilon_{klm} w_a$$

$$\frac{d\hat{e}^k}{dt} = w_{kl} \hat{e}^l$$

$$\hat{e}^k(t+dt) = (\delta_{kl} + w_{kl} dt) \hat{e}^l(t)$$

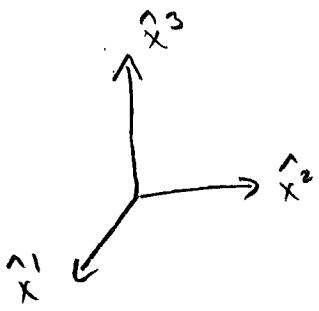
$$(L^a)^{mn} = i \epsilon^{man}$$

$$\begin{aligned}&= (\underbrace{\delta_{kl} + \epsilon_{klm} w_a dt}_n) \hat{e}^l(t) \\ &\quad + i w_a dt (L^a)^{kl}\end{aligned}$$

$$= \underbrace{e^{i w_a dt L^a}}_{\text{passive}} \hat{e}^l(t)$$

Recall  $R = e^{-i \phi \hat{n} \cdot \vec{\ell}}$  = active rotat

$e^{i \phi \hat{n} \cdot \vec{\ell}}$  = passive rotat



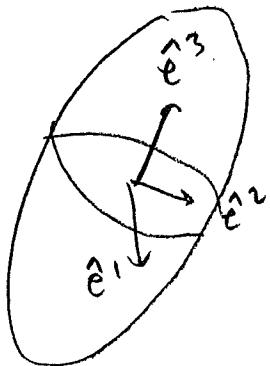
Let  $\hat{x}^m$  be Cartesian unit vectors  
of an inertial reference frame  
"space fixed frame"

$$\text{orthonormal: } \hat{x}^m \cdot \hat{x}^n = \delta_{mn}$$

$$\text{Arbitrary vector } \vec{A} = \sum_m A_m \hat{x}^m$$

$$A_m = \hat{x}^m \cdot \vec{A} = \hat{x}^m \cdot \sum_n A_n \hat{x}^n = \sum_n A_n \delta_{mn} = A_m$$


---



Let  $\hat{e}^k$  be principal axes of a rigid body

$$\text{orthonormal } \hat{e}^k \cdot \hat{e}^l = \delta_{kl}$$

These can be taken as basis vectors of a  
coordinate system attached to object

"body fixed frame"

Not an inertial reference frame because object rotates  
+ so does the body fixed frame.

Advantage: Inertia tensor is diagonal in body fixed frame  
+ also doesn't change in time  
(object not changing wrt. body fixed frame)

$$\text{Arbitrary vector } \vec{A} = \sum_k A'_k \hat{e}^k$$

$$A'_k = \hat{e}^k \cdot \vec{A}$$

use primes now for  
components in  
body fixed frame  
[Marion & Thornton, p37]

Consider was another notation than prime, might tilde?

Let  $\hat{e}_m^k$  ( $m=1, 2, 3$ ) be component of  $\hat{e}^k$  in space fixed frame

$$\hat{e}^k = \sum_m \hat{e}_m^k \hat{x}^m \quad \hat{e}_m^k = \hat{e}^k \cdot \hat{x}^m$$

We can use  $\hat{e}_m^k$  to relate  $A_m$  and  $A'_k$

$$A'_k = \hat{e}^k \cdot \vec{A} = \sum_m \hat{e}_m^k \hat{x}^m \cdot \vec{A} = \sum_m \hat{e}_m^k A_m^m$$

Define matrix  $R$  by matrix of elements  $R_{km} \equiv \hat{e}_m^k$   
↑ row ↑ column

$$A'_k = \sum R_{km} A_m^m \quad \begin{pmatrix} A'_1 \\ A'_2 \\ A'_3 \end{pmatrix} = R \begin{pmatrix} A_1 \\ A_2 \\ A_3 \end{pmatrix}$$

if  $R$  have = transpose  
 of  $R$  and the others  
 transpose

[NB - passive transformation]

Also

$$A_m = \hat{x}^m \cdot \vec{A} = \hat{x}^m \cdot \sum_k A'_k \hat{e}^k = \sum_k \hat{e}_m^k A'_k$$

$$= \sum R_{mk} A'_k = \sum R_{mk}^T A'_k \quad \begin{pmatrix} A_1 \\ A_2 \\ A_3 \end{pmatrix} = R^T \begin{pmatrix} A'_1 \\ A'_2 \\ A'_3 \end{pmatrix}$$

(Because  $\det R = \pm 1$   
 but right handed  
 and so are  
 correspondingly  
 correlated)

ie  $R$  is an orthogonal matrix.

will later characterise  
 of Euler angles

⇒ the transformation from  $\hat{x}^m$  to  $\hat{e}^k$  is a rotation.

If both  $\hat{x}^m$  &  $\hat{e}^k$  are right handed, then  $\det R = 1$

Group of orthogonal matrices of determinant 1 is called  $SO(3)$

[NB: do not introduce summation convention because apparently violated by  $I_k e_m^k$ !]

53

$$\text{Recall: } \vec{L} = \vec{I} \cdot \vec{\omega}$$

$$\text{In space-fixed frame: } L_m = \sum_n I_{mn} \omega_n$$

$$\text{In body-fixed frame: } L'_k = \sum_l I'_{kl} \omega'_l \quad (*)$$

We know  $L'_k = \sum_m R_{km} L_m$  and  $\omega'_k = \sum_m R_{km} \omega_m$ . Also  $\omega'_n = \sum_l R_{ln} \omega'_l$   
because  $\vec{L} + \vec{\omega}$  are vectors ( $1^{\text{st}}$  rank tensors)

How are  $I'_{kl}$  and  $I_{mn}$  related?

$$L'_k = \sum_m R_{km} L_m = \sum_m \sum_n R_{km} I_{mn} \omega_n = \sum_m \sum_n \sum_l R_{km} I_{mn} R_{ln} \omega'_l$$

Comparing w/ (\*) we see

$$I'_{kl} = \sum_m \sum_n R_{km} R_{ln} I_{mn}$$

$I$  transforms as a  $2^{\text{nd}}$  rank tensor

E.g. a 3<sup>rd</sup> rank tensor  $S_{mnp}$  transforms as

$$S'_{kli} = \sum_m \sum_n \sum_p R_{km} R_{ln} R_{ip} S_{mnp}$$

Compute inertial term in body fixed frame

$$I'_{kl} = \sum_m \sum_n R_{km} R_{ln} I_{mn} \quad (*)$$

$$= \sum_m \sum_n e^k{}_m I_{mn} e^l{}_n$$

Recall  $\sum_n I_{mn} e^l{}_n = I_k e^k{}_m$  defines principal axes & moments

$$I'_{kl} = \sum_m e^k{}_m I_m e^l{}_m$$

$$= I_h \hat{e}^k \cdot \hat{e}^l$$

$$= I_h \delta_{kl}$$

$$\overset{\leftrightarrow}{I}' = \begin{pmatrix} I_1 & 0 & 0 \\ 0 & I_2 & 0 \\ 0 & 0 & I_3 \end{pmatrix} \quad \begin{array}{l} \text{diagonal in body-fixed frame} \\ [\text{make sense because object balanced w.r.t. principal axes}] \end{array}$$

Can write

$$\overset{\leftrightarrow}{I}' = R \overset{\leftrightarrow}{I} R^{-1} \quad \begin{array}{l} \leftarrow \text{similarity transformation} \\ [\text{check this agrees w/ (*)}] \end{array}$$

Thm: any symmetric real matrix can be diagonalized by an orthogonal similarity fx

[also: any hermitian ... unitary]

Summarize

$$\vec{L} = \vec{I} \cdot \vec{\omega} \quad \xrightarrow{\text{space fixed}} \quad L_m = \sum_n I_{mn} \omega_n$$

$$\begin{aligned} \vec{L}' &= R \vec{L} = R \vec{I} \cdot \vec{\omega} \\ &= R \vec{I} R^{-1} R \vec{\omega} \quad \xrightarrow{\text{body fixed}} \quad L'_k = I_k \omega'_k \\ &= \vec{I}' \cdot \vec{\omega}' \end{aligned}$$

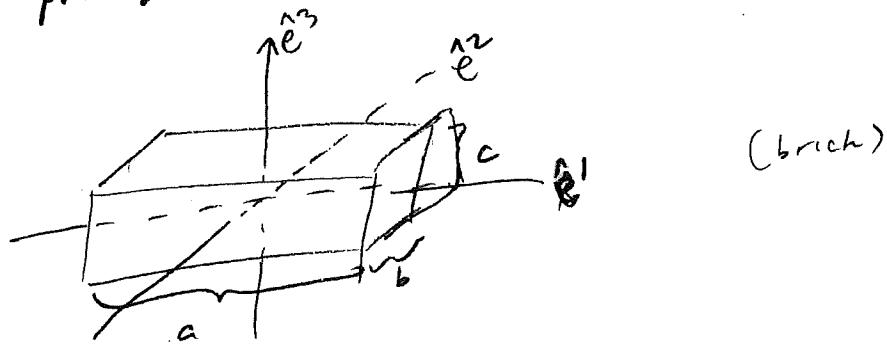
$$T = \frac{1}{2} \vec{\omega} \cdot \vec{I} \cdot \vec{\omega} \quad \longrightarrow \quad T = \frac{1}{2} \sum_m \sum_n \omega_m I_{mn} \omega_n$$

$$\begin{aligned} &= \frac{1}{2} \vec{\omega} R R^{-1} \vec{I} R^{-1} R \vec{\omega} \\ &= \frac{1}{2} \vec{\omega}' \cdot \vec{I}' \cdot \vec{\omega}' \quad \longrightarrow \quad T = \frac{1}{2} \sum_k I_k \omega'_k^2 \end{aligned}$$

$$\left( \begin{array}{l} \vec{\omega}' = R \vec{\omega} \\ \vec{\omega}'^T = \vec{\omega}^T R^T = \vec{\omega}^T R^{-1} \end{array} \right)$$

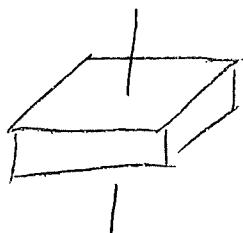
SK  
SK  
SL

) If object has some obvious symmetry  
principal axis should be obvious



$$I = \frac{1}{12} M \begin{bmatrix} b^2 + c^2 & 0 & 0 \\ 0 & c^2 + a^2 & 0 \\ 0 & 0 & a^2 + b^2 \end{bmatrix}$$

O If  $a = b$ :



(square plate)

$$I = \frac{1}{12} M \begin{bmatrix} a^2 + c^2 & 0 & 0 \\ 0 & a^2 + c^2 & 0 \\ 0 & 0 & 2a^2 \end{bmatrix}$$

$I_1 = I_2 \Rightarrow$  not only  $\hat{e}^1 + \hat{e}^2$  but  
any linear comb. thereof is a principle axis

$$\begin{aligned} I \cdot (\alpha \hat{e}^1 + \beta \hat{e}^2) &= \alpha \overset{\leftrightarrow}{I} \cdot \hat{e}^1 + \beta \overset{\leftrightarrow}{I} \cdot \hat{e}^2 = \alpha I_{1,1} \hat{e}^1 + \beta I_{2,2} \hat{e}^2 \\ &= I_{1,1} (\alpha \hat{e}^1 + \beta \hat{e}^2) \end{aligned}$$

(Any axis in  $xy$  plane is a principle axis)  
(this is why prop of orthogonality fails)

Degrad eigenvalue  $\Rightarrow$  any linear comb. of corresponding  
eigenvectors is an eigenvector of same eigenvalue

SF  
88  
57

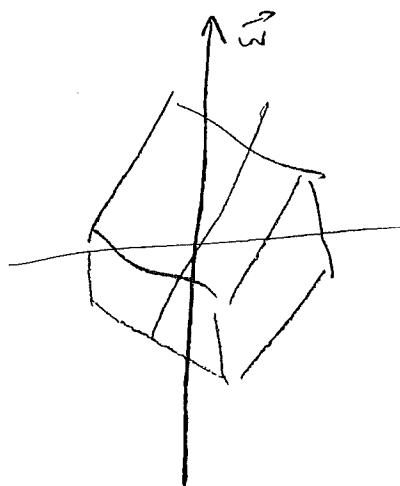
For a cube       $\overset{\leftarrow}{I} = \frac{1}{6} m a^2 \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$

$$I_x = I_y = I_z$$

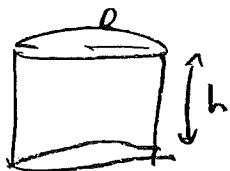
$\downarrow$   
 $\Rightarrow$  any vector is an eigenvector

$$\overset{\rightarrow}{L} = \overset{\leftarrow}{I} \cdot \overset{\rightarrow}{\omega} = \frac{1}{6} m a^2 \overset{\rightarrow}{\omega} \text{ for any } \overset{\rightarrow}{\omega} \text{ through center of cube}$$

$\Rightarrow$  rotation about any axis is dynamically stable  
 [see ahead]



cylinder:



For what value of  $h/R$   
 is cylinder dynamically stable  
 about any axis?

[ask next class]

$$\frac{1}{2} m R^2 = \frac{1}{2} m h^2 + \frac{1}{4} m R^2$$

$$\Rightarrow h = \sqrt{3} R$$